IN THE CLAIMS

1. (currently amended) A method for operating a gas turbine engine, said gas turbine engine comprising a nozzle including a plurality of chevrons coupled to the nozzle, said method comprising:

positioning a plurality of tubes azimuthally around an outer periphery of the nozzle;

coupling an upstream end of each of the plurality of tubes to a manifold;

coupling a downstream end of each of the plurality of tubes to the nozzle such that the plurality of tubes each externally extend away from the manifold,

channeling compressed air from the gas turbine engine to a noise suppression system that includes ~~a~~ the manifold and ~~a~~ the plurality of tubes ~~coupled to the manifold~~; and

selectively operating the noise suppression system such that air discharged from the noise suppression system enhances a streamwise vortex generated downstream from each respective chevron.

2. (original) A method in accordance with Claim 1 wherein selectively operating the noise suppression system further comprises selectively operating the noise suppression system such that air discharged from the noise suppression system facilitates reducing gas turbine noise generated during engine operation.

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3. (previously presented) A method in accordance with Claim 1 wherein channeling compressed air from the gas turbine engine to a noise suppression system further comprises:

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channeling compressed air from the gas turbine engine into the manifold; and

discharging the air from the manifold into the respective chevron flowpath through the plurality of tubes.